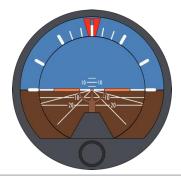


## Inertial Labs

The **Inertial Labs Attitude and Heading Reference Systems, AHRS-II** is the next generation of enhanced, high-performance strapdown systems that determines absolute orientation (**Heading, Pitch and Roll**) for any device on which it is mounted. Orientation is determined with high accuracy for both motionless and dynamic applications.



The Inertial Labs AHRS-II-P utilizes Tactical-grade IMU: 3-axes each of precision accelerometers & gyroscopes and Ultra High precision, gyro-compensated, embedded Fluxgate Compass to provide accurate Heading/Yaw, Pitch and Roll of the device under measure.

Integration of gyroscopes' output provides high frequency, real-time measurement of the device rotation about all three rotational axes. Accelerometers and Fluxgate magnetometer measure absolute Pitch, Roll and magnetic Azimuth at AHRS initial alignment as well as providing ongoing corrections to gyroscopes during operation.

#### **KEY FEATURES AND FUNCTIONALITY**

- State-of-the-art Kalman Filter based algorithms for different dynamic motions of UAV, Aircrafts, Helicopters, Vessels, Ships, ROV, UUV, Robots, UGV, AGV, Gimbals and Antennas
- Highly accurate, sensitive, and temperature stable Fluxgate magnetometers (in-house technology)
- · Input from External GNSS Receiver;
- 0.6 deg Gyro-Stabilized Heading and 0.08 deg Pitch & Roll Dynamic Accuracy over operational temperature range
- 1 deg/hr gyroscopes Bias in-run stability
- 5 µg accelerometers Bias in-run stability
- Suitable for Primary Attitude Reference and as Motion Control System for marine applications
- Embedded 2D and 3D magnetic calibration on hard and soft iron
- RS-232, RS-422, and CAN 2.0 interfaces
- Full temperature calibration of all sensing elements
- Up to 200 Hz data update rate (AHRS) and 2000 Hz data rate (IMU)
- Compact design. Environmentally sealed (IP67)

One of the key elements to the success of Inertial Labs AHRS-II is its use of **Inertial Labs 8mm Fluxgate Magnetometers,** which have distinct advantages over commonly used magneto-inductive or magneto-resistive magnetometers. In operation over time and temperature fluxgate magnetometers have superior stability and repeatability. In terms of sensitivity, fluxgate magnetometers provide up to two orders of magnitude increased sensitivity.

In addition to the performance advantages, unlike the chip-level magnetometer technology, fluxgate magnetometer technology has been depended on for over 70 years to provide an accurate reference to North. It remains the most reliable magnetic sensor technology for determining an object's heading.

### **AHRS-II-P** key specifications summary

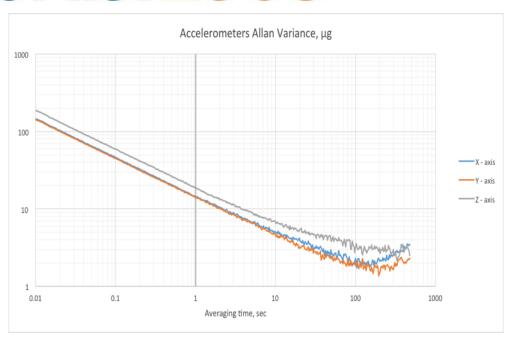
Parameter	Units	
Status		In production (Since 2006)
Heading static accuracy	Heading static accuracy deg 0.3	
Heading dynamic accuracy	deg	0.6 (gyro magnetic)
Heading dynamic accuracy	deg	0.1 (with GNSS input)
Pitch & Roll static accuracy	deg	0.05
Pitch & Roll dynamic accuracy	deg	0.08
Gyroscopes Bias in-run stability	deg/hr	1
Accelerometers Bias in-run stability	mg	0.005

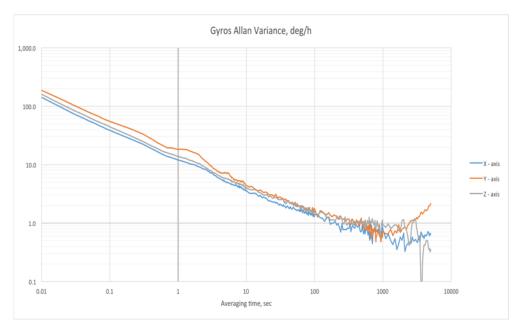


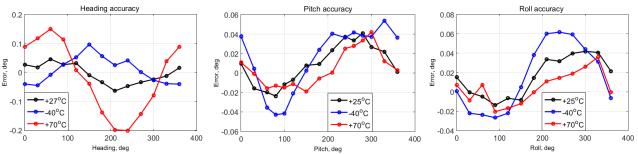
### **Inertial Labs AHRS-II-P key performance**

Parameter Status	Units	AHRS-II-P In production			
Status	Without GNSS	Euler angles; Quaternion; Relative Altitude; IMU data (angular rates,			
Output signals	input With GNSS input	accelerations); Magnetic field; Pressure; Delta Theta, Delta Velocity Horizontal & Vertical Positions (LAT, LONG); Heading, Pitch, Roll, Velocity, PPS time, IMU data (angular rates, accelerations)			
Input signals		External GNSS receiver, External magnetic compass			
Update rate (AHRS data)	Hz	1 200 (user settable)			
Update rate (IMU data)	Hz	2000			
Start-up time	sec	< 1			
Full Accuracy Data (Warm-up Time)	sec	10			
Heading	Units				
Range	deg	0 to 360			
Angular Resolution	deg	0.01			
Static Accuracy Over Operational Temperature Range	deg	0.3			
Dynamic Accuracy Over Operational Temperature Range	deg RMS	0.6			
Pitch and Roll	Units	100 1100			
Range: Pitch, Roll	deg	±90, ±180			
Angular Resolution	deg	0.01			
Static Accuracy in whole Temperature Range Dynamic Accuracy Over Operational Temperature Range	deg deg RMS	0.05 0.08			
Relative altitude	Units	0.08			
Measurement range	meters	-600 to 9000			
Resolution	meters	0.01			
Relative accuracy	meters	<1			
Gyroscopes	Units	1			
Measurement range	deg/sec	±450 / ±950			
Bandwidth	Hz	200			
Bias in-run stability (RMS, Allan Variance)	deg/hr	1			
Bias residual error over operational temperature range	deg/hr, RMS	<35			
SF accuracy	ppm	3000			
Noise (Angular Random Walk)	deg/√hr	<0.2			
Non-linearity	ppm	100			
Axis misalignment	mrad	0.15			
Accelerometers	Units				
Measurement range	g	±8 / ±15			
Bandwidth	Hz	200			
Bias in-run stability (RMS, Allan Variance)	mg	0.005			
Bias residual error in temperature range, RMS	mg	<0.5			
SF accuracy	ppm	100			
Noise (Velocity Random Walk)	m/s/√hr	0.015 500			
Non-linearity Axis misalignment	ppm	0.1 mrad			
Magnetometers (Embedded Fluxgate Compass)	mrad Units	U.1 IIIIdu			
Measurement range	Gauss	±1.6			
Bias in-run stability, RMS	nT	0.2			
Noise density, PSD	nT√Hz	0.3			
SF accuracy	%	0.02			
Bandwidth	Hz	50			
Pressure	Units				
Measurement range	hPa	300 – 1100			
Bandwidth	Hz	50			
Resolution	Pa	2			
Bias in-run stability (RMS, Allan Variance)	Pa	0.8			
Noise density	Pa/√Hz	1			
Environment	Units				
Operating temperature	deg C	-40 to +70			
Storage temperature	deg C	-50 to +85			
MTBF (G <sub>M</sub> @ +65degC)	hours	100,000			
Vibration & Shock		MIL-STD-810G			
EMC/EMI	Huite	MIL-STD-461			
Electrical Supply voltage	Units	0 to 26			
Supply voltage	V DC Watts	9 to 36 1.4			
Power consumption Output Interface	- vvatts	1.4 RS-232, RS-422, CAN 2.0			
Output Interrace Output data format	<u>-</u>	Binary, NMEA 0183 ASCII characters			
Physical Physical	Units	Dinary, WHEN 0103 ASOII Characters			
Size	mm	120 x 50 x 53			
Weight	gram	280			
Weight	J. WIII				

# **Inertial Labs**

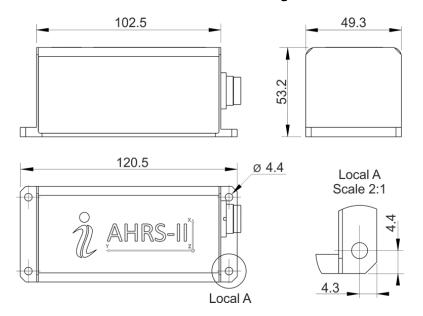








### AHRS-II-P mechanical interface drawing



### Notes:

- All dimensions are in millimeters. 1.
- All dimensions within this drawing are subject to change without notice. Customers should obtain final drawings before designing any interface hardware.

Connector type: Binder Series 723. Male receptacle, 24 pin, shielded, rearmounting

AHRS-II-P part numbers description										
Model	Gyro	Accel	Calibration	Connector	Color	Data Logger	Version	Interface		
AHRS-II-P	G450	A8	TMGA	C3 (default)	B (default)	S8 (option)	V1	1		
	G950	A15			D	S64 (option)		2		
A40			G			4				
			W			5				
						11				
								22		
Example: AHRS-II-P-G450-A8-TMGA-C3B-V1.11	11					145				
						245				

#### Description:

- AHRS-II-P: Professional Version of Attitude and Heading Reference System
- G450: Gyroscopes measurment range =  $\pm 450$  deg/sec G950: Gyroscopes measurment range =  $\pm 950$  deg/sec
- A8: Accelerometers measurement range = ±8 g
- A15: Accelerometers measurement range = ±15 g
- A40: Accelerometers measurement range = ±40 g
- TMGA: Magnetometers, Gyroscopes and Accelerometers C3: 24 pins connector (RS-232, RS-422, CAN, Ethernet interfaces)
- B Black Color (default)
- D Desert Color (Desert tan, color code 33446 (tan 686A) per FED-STD-595, Change Notice 1.)
- G Green
- W White
- S8: 8GB embedded Data Logger (optional)
- S64: 64GB embedded Data Logger (optional)
- VX.1: RS-232 interface
- VX.2: RS-422 interface
- VX.4: CAN interface VX.5: Ethernet interface
- VX.11: two RS-232 interfaces
- VX.22: two RS-422 interfaces
- VX.145: RS-232, CAN and Ethernet interfaces VX.245: RS-422, CAN and Ethernet interfaces